PCW = Previously Complied With

## N/A = Not Applicable

See MX Records for further Petails

## Summary Compliance Report For Aircraft Registration N758QV

© 2025 Veryon

Airframe I Textron Aviation Inc.	I P172K I Part No	Serial No. R1723284	Position: n/a	TSN:	CSN:
Ainfrance   Toytron Aviation Inc.	IKITEN I Fait No	Colidi Hours It is money.			

4 2025 AFTT 2316.0

As of -

Issue No. SB No. CAA No. Effective Date	Description	CM	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 2024-26-02 SB# CAA# 3/7/2025	The FAA is Issuing this AD to Inspect for Cracks and Thickness of the Buckle Handle. The Unsafe Condition, Could Prevent a Strap from Releasing When the Buckle is Rotated, Which Could Result in Occupants Not Being Able to Release the Buckle cont'd	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: #55 371763
ssue# 2024-14-03 SB# CAA# 8/20/2024	To Address Autopilot Software That Does Not Properly Handle Certain Hardware Failures of the Primary Pitch Servo. The Unsafe Condition, if Not Addressed, Could Result in Un-Commanded Automatic Pitch Trim Runaway and Loss of Control of the Airplane	Hrs: C:	- NA	No	Hrs: C:	1 2 3 4 Signed: 185 3711763
ue# 2023-10-02 SB# CAA#	To Address Radio Altimeter Anomalies That Are Undetected by the Automation or Pilot, Particularly Close to the Ground (e.g., Landing Flare), Could Lead to Loss of Continued Safe Flight and Landing. Additionally, Radio Altimeter Anomalies Could, contd.	Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: 185 370763

Default

Content Revision: 4/3/2025

Report contents limited by: FAA ADs

Report created on 4/4/2025 by a.james@ajaviationservices.com

Page 1 of 12

Issue No. SB No. CAA No Effective Date	Description	CAV	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No 4. Authorized By
Issue# 2002-26-03 SB# CAA# 2/18/2003	To detect & correct incorrect installation of the air filter, which could result in failure of the air filter	 Hrs: C:	- N/A	No	Hrs:	1 2 3 4 Signed: #8- 3711763
Issue# 96-09-06 SB# CAA# 6/7/1996	[Recurring] TO PREVENT GASKET PARTICLES FROM ENTERING THE CARBURETOR BECAUSE OF AIR FILTER GASKET FAILURE, WHICH COULD, CONTD.	 Hrs: C:	Replace filter element — tracked in QMX	Yes	Hrs: C:	1 2 3 4 Signed: A8- 3711763
Issue# 81-15-03 SB# CAA# 7/20/1981	TO PREVENT POSSIBLE FAILURE OF THE ALUMINUM AIR FILTER RETAINER SCREEN OR GASKETS WITH POTENTIAL, CONTD.	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: #8- 371763

Issue No. SB No. CAA No. Effective Date	Description	CW	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 2006-24-07 SB# CAA# 1/3/2007	To detect potentially unsafe conditions that could result in a propeller blade separating from the hub, contd.	Hrs: C:	- N/A	No	Hrs: G:	1 2 3 4 Signed: #85 371763
Issue# 2005-14-11 SB# CAA# 8/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: AS 3211763
SB#	To prevent propeller failure of the propellers returned to service by BASCO, & possible loss of airplane control	Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: <u>\$5</u>
# 94-17-13 P SB# H AA# C	POSSIBLE PROPELLER	 C:	- N/A	Yes	Hrs: C:	1 2 3 4 Signed: AS

			Method of Compliance/Applicability	Recurring	Next Due	3. Cert. No. 4. Authorized By
Issue# 91-19-03 SB# CAA# 9/29/1991	TO PREVENT OPERATION WITH COLLAPSED OIL FILTER ELEMENTS, WHICH CAN RESULT IN LOSS OF OIL PRESSURE, CONTD.	 Hrs: C:	- N/A	No	Hrs:	1 2 3 4 Signed: #8 3711763
Issue# 88-03-06 SB# CAA# 4/15/1988	TO PREVENT POSSIBLE LOSS OF ENGINE OIL AND SUBSEQUENT ENGINE FAILURE	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: #85 3717763
R1 SB#	TO PREVENT OIL PUMP DRIVE SHAFT FAILURE AND SUBSEQUENT ENGINE FAILURE DUE TO OIL PRESSURE LOSS	 Hrs: C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: 185 371763

Issue No. SB No. CAA No. Effective Date	Description	CM	Amendment No.  Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 2007-26-09 SB# CAA# 1/30/2008	To prevent failure of the propeller blade from fatigue cracks in the aluminum blade shank radius, which can, contd.	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: <del>AS</del> 371763

Issue No. SB No. CAA No. Effective Date	Description	CNV	Amendment No. Method of Compliance/Applicability	Recurring	Next Due 3	Facility Cert Type Cert No Authorized By
Issue# 2000-23- SB# CAA# 12/12/2000	To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd.	 Hrs: C:	- NA	No	Hrs: G:	1 2 3 4 Signed: 185 371763
Issue# 98-19-02 SB# CAA# 11/9/1998	To prevent a piston pin failure from causing secondary engine damage that results in loss of oil or total, contd.	 Hrs: C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: #85 371763
Issue# 97-26-17 C SB# CAA#	[Recurring] TO PREVENT CRANKSHAFT FAILURE AND SUBSEQUENT ENGINE FAILURE	 Hrs: C:	- PCW	Yes	 Hrs: C:	1 2 3 4 Signed: #8 371763
ssue# 93-10-02 SR#	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY	 Hrs: C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: <del>AS</del> 371763
ue# 92-04-09 F	TO PREVENT ENGINE FAILURE CAUSED BY FAILURE OF THE ROCKER SHAFT HOLD DOWN STUD	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: 185 37176

Content Revision: 4/3/2025

Report contents limited by: FAA ADs

Report created on 4/4/2025 by a.james@ajaviationservices.com

Page 9 of 12

Issue No. SB No. CAA No. Effective Date	Description	CAV	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 2023-05-16 SB# CAA# 3/15/2023	To Prevent Departure of Counterweight and Retaining Hardware from the Crankshaft Assembly. The Unsafe Condition, if Not Addressed, Could Result in Loss of Engine Oil Pressure, Catastrophic Engine Damage, Engine Seizure, and Consequent Loss of the Aircraft	C: -	NA	No	Hrs: C:	1 2 3 4 Signed: #5 3711763
Issue# 2012-03-06 C SB# CAA#	To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and damage to the airplane	 Hrs: C:	- N/A	No	 Hrs: C:	1 2 3 4 Signed: #8 371763
sue# 2011-26-07	To prevent engine failure and loss of control of the airplane due to migration of the magneto impulse, contd.	 Hrs: C:	- NA	No	 Hrs: C:	1 2 3 4 Signed: #85 371763
e# 2010-11-04 h SB# CAA# P	To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane	Hrs: C:	- ? C W	No	 Hrs: C:	1 2 3 4 Signed: AS 371176

Issue No. SB No. CAA No. Effective Date	Description	CM	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 71-22-02 R(1) SB# CAA#	[Recurring] TO DECREASE THE POSSIBILITY OF FAILURE OF THE NOSE GEAR STRUCTURE	Hrs: C:	- N/A	Yes	Hrs: C:	1 2 3 4 Signed: #85 371763
ssue# 69-15-03 SB# CAA# 8/20/1969	[Recurring] TO DETECT CRACKS IN THE MUFFLER ASSEMBLY	 Hrs: C:	- N/A	Yes	Hrs: C:	1 2 3 4 Signed: <del>ASS</del> 371763
ue# 68-17-04 SB# CAA#	[Recurring] TO ASSURE PROPER OPERATION OF THE STALL WARNING SYSTEM IN FLIGHT	 Hrs: C:	- N/A	Yes	Hrs: C:	1 2 3 4 Signed: A85 3711763

Issue No. SB No. CAA No. Effective Date	Description	C/W	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
	To Prevent Loss of Engine Power. The Unsafe Condition, if Not Addressed, Could Result in Failure of the Engine, In-Flight Shutdown, and Loss of Control of the Airplane	 Hrs: C:	- NA	No	Hrs: C:	1 2 3 4 Signed: 185 371763

Content Revision: 4/3/2025

Report contents limited by: FAA ADs

Report created on 4/4/2025 by a.james@ajaviationservices.com

Page 7 of 12

Issue No. SB No. CAA No. Effective Date	Description	CAV	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 79-08-03 C SB# CAA#	TO PREVENT ELECTRICAL SYSTEM FAILURE, SMOKE IN THE COCKPIT, AND/OR FIRE IN THE WIRE BUNDLE BEHIND THE INSTRUMENT PANEL	 Hrs: C:	- N/A	No	Hrs: C:	1. — 2. — 3. — 4. — Signed: 155 3711763
Issue# 77-12-08 SB# CAA#	TO PREVENT UNWANTED PROPELLER ROTATION	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: #85 3711763
Issue# 77-02-09 SB# CAA#	WING FLAP ACTUATOR DATE CODE STAMP	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: 455 370763
sue# 74-06-02 SB# CAA#	[Recurring] TO PREVENT POSSIBLE LEAKAGE OF CARBON MONOXIDE INTO THE CABIN HEATER SYSTEM	 C:	- W/A	Yes	Hrs: C:	1 2 3 4 Signed: #55 371763
ue# 73-17-01 F	TO ADVISE THE PILOT CONCERNING PROPER FUEL TRANSFER PUMP OPERATION	 Hrs: C:	- NA	No	Hrs: C:	3

Content Revision: 4/3/2025

Report contents limited by: FAA ADs

Report created on 4/4/2025 by a.james@ajaviationservices.com

Issue No. SB No. CAA No. Effective Date	Description	CW	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 83-10-0 SB# CAA# 6/30/1983	TO PREVENT POSSIBLE JAMMING OF THE ELEVATOR CONTROL	Hrs: C:	- PCW	No	Hrs: C: -	1 2 3 4 Signed: #8- 371763
Issue# 81-16-09 SB# CAA# 8/13/1981	TO ENSURE THE INTEGRITY OF THE ELEVATOR CONTROL SYSTEM	 Hrs: C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: #5- 3711763
ssue# 81-05-01 SB# CAA# 3/2/1981	TO REDUCE THE POSSIBILITY OF FUEL DEPLETION DUE TO INCORRECT FUEL QUANTITY MARKINGS	 Hrs: C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: AS 3711763
SB# CAA#	TO ASSURE CONTINUED STRUCTURAL INTEGRITY OF THE WING FLAP DIRECT CABLE, THEREBY PREVENTING POSSIBLE, CONTD.	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: AS 3711763
# 80-07-03 SB# S AA# O	O PRECLUDE FAILURE OF THE ENGINE OIL PRESSURE PUMP DRIVE HAFT AND RESULTING IL PRESSURE LOSS, ONTD.	 Hrs: C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: AS 371176

Issue No. SB No. CAA No. Effective Date	Description	CNV	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 97-01-13 SB# CAA# 2/3/1997	TO PREVENT FUEL, OIL, OR HYDRAULIC SYSTEMS FAILURE CAUSED BY A COLLAPSED HOSE	 Hrs: C:	- PCW	No	Hrs: C:	2 3 4 Signed: #85 3711763
Issue# 96-12-22 SB# CAA# 7/31/1996	[Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	 Hrs: C:	- N/A	Yes	Hrs: C:	1 2 3 4  Signed: #8- 311763
Issue# 86-24-07 SB#	TO PREVENT ENGINE POWER INTERRUPTION DUE TO LOSS OF ATTACHMENT OF THE ENGINE CONTROLS	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: 185 371763
ssue# 86-19-11	TO ELIMINATE THE POSSIBILITY OF ENGINE POWER REDUCTION DUE TO CONTAMINATED FUEL	 Hrs: C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: #85 -371763
sue# 83-22-06	TO PREVENT POSSIBLE LOSS OF AN AILERON HINGE PIN	 Hrs: C:	- SCM	No	Hrs: C:	1 2 3 4 Signed: <del>185</del> 3711763

Issue No. SB No. CAA No. Effective Date	Description	CAV	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 2008-10-02 SB# CAA# 5/12/2008	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.		- N/A	No	Hrs: C:	1 2 3 4 Signed: 15- 3711763
Issue# 2008-02-18 SB# CAA# 2/28/2008	To prevent premature separation of the collar, which could result in the parachute failing to, contd.	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: <del>185</del> 3711763
Issue# 2001-23-03 SB# CAA# 12/27/2001	[Recurring] To detect and correct any chafing between the map light switch and the bordering fuel line, which could, contd.	 Hrs: C:	C/W by inspection (insulator) per SEB00-1	Yes	Hrs: C:	1 2 3 4 Signed: AS 371763
Issue# 2000-06-01 SB# CAA# 5/5/2000	To prevent foreign material from entering the fuel system and engine, which could result in loss of engine power, contd.	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: #85
Issue# 99-27-02 SB#	Fuel selector valve cams or fuel selector valves, when installed, could result in an inadequate supply of, contd.	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: AS 3711763

Content Revision: 4/3/2025

Report contents limited by: FAA ADs

Report created on 4/4/2025 by a.james@ajaviationservices.com

Page 3 of 12

Issue No. SB No. CAA No. Effective Date	Description	CW	Amendment No. Method of Compliance/Applicability	Recurring	Next Due	1. Facility 2. Cert. Type 3. Cert. No. 4. Authorized By
Issue# 2022-03-15 SB# CAA# 3/21/2022	To Ensure That the Amount of Fuel Indicated is the Amount of Fuel Available. The Unsafe Condition, if Not Addressed, Could Result in Fuel Starvation and Engine Shutdown Which Could Result in the Inability to Arrive at the Destination Airport or, contd.	 Hrs: C:	- N/A	No	Hrs: C:	1 2 3 4 Signed: 485 371763
Issue# 2020-18-01 SB# CAA# 11/12/2020	[Recurring] To Detect and Address Cracking of the Wing Strut Attach Point. The Unsafe Condition, if Not Addressed, Could Result in Failure of the Wing in Operation, Which Could Result in Loss of Control of the Airplane	Hrs: C:	Due at AFTT 4000	Yes	 Hrs: C:	1 2 3 4 Signed: ASS 3711763
Issue# 2011-10-09 SB# CAA# 6/17/2011	[Recurring] To prevent seat slippage or the seat roller housing from departing the seat rail, which may consequently cause, contd.	 Hrs: C:	Inspection Due annually	Yes	Hrs: C:	1 2 3 4 Signed: A 3711763
Issue# 2008-26-10 C SB# CAA#	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	 C:	- PCW	No	Hrs: C:	1 2 3 4 Signed: ASS 3711763